

67,008-079
Sikorsky Docket No.:S-5696**AMENDMENTS TO THE SPECIFICATION:**

Please replace the following numbered paragraphs with the following rewritten paragraphs:

- [17] Referring to Figure 2, a rotor blade 20 (only one illustrated) of the rotor assembly 12 includes an inboard section 22, an intermediate section 24, and an outboard section 26. The inboard, intermediate, and outboard sections 22, 24, 26 define the span of the main rotor blade 20. The blade sections 22, 24, 26 define a blade radius R between the axis of rotation A and a blade tip 28. The blade tip 28 is the distal end of the rotor blade 20.
- [19] A vortex generator 30 is located adjacent the blade tip 28. The vortex generator preferably includes a multiple of vorticity generating protuberances 32 which extend generally parallel to a longitudinal blade axis P. The longitudinal blade axis P is preferably a feathering axis of the rotor blade 20. The vorticity generating protuberances 32' are alternatively or additionally selectively deployable (Figure 3). That is, the vorticity generating protuberances 32' may alternatively or additionally be actively deployable related the azimuthally position.